



# A General Purpose Telemetry Monitor (GPTM) for the Hubble Space Telescope (HST)

Repurposing an Existing Design

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# Introduction

- Hubble Space Telescope (HST) operating for 21 years with 24x7 ground support
- A proven design was modified to provide an autonomous telemetry monitor
- There are clear benefits to design reuse
- There are also challenges and pitfalls to avoid

First, some background

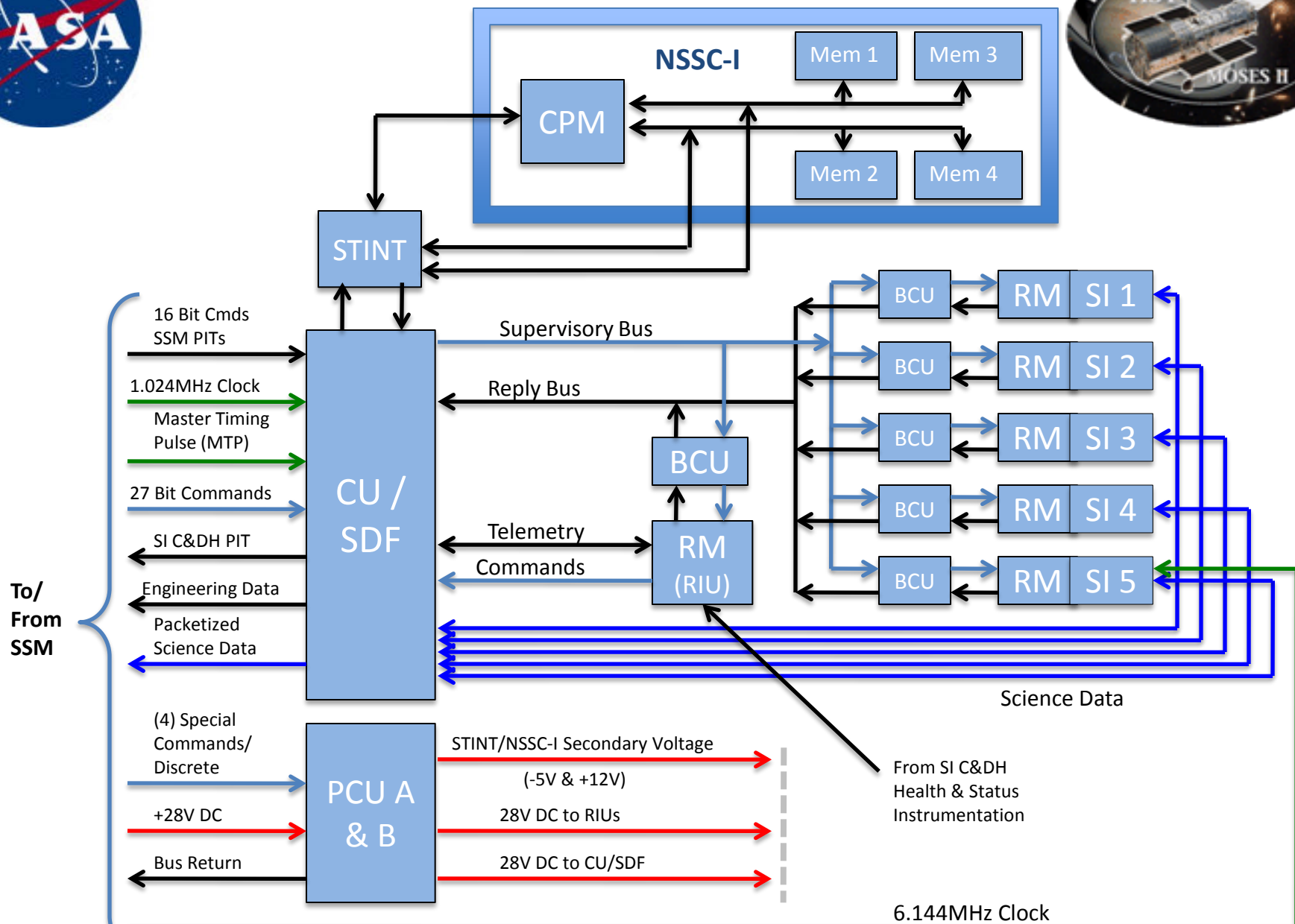


# Background

- HST is a mix of old and new technology
  - New Science Instruments are state of the art
  - The NASA Standard Spacecraft Computer (NSSC-1) designed in mid-1970s, first flown in 1980
    - NSSC-1 is part of the Science Instrument Command & Data Handling (SI C&DH) subsystem
    - On-orbit SI C&DH replaced with “new” 20 year old SI C&DH during Servicing Mission 4 (May 2009)
    - Programmed in NSSC-1 Assembly Language (18-bit octal)
      - Many executive programs inherited with little change from earlier NSSC-1 based spacecraft, most notably Solar Maximum Mission (SMM)



# SI C&DH Hardware





# Automation Problem



- Transition from fully manned to autonomous lights out operations
  - HST was continuously monitored from the ground for 21 years
  - Flight operations staffing reduced to 8x5 support on June 13, 2011
    - Planning began before the last servicing mission
    - Need for an onboard autonomous anomaly response was recognized
      - » Minimize time between anomaly and response
      - » Avoid Flight Operations Team involvement in routine “anomalies”
      - » Reduce dependence on downlinked telemetry for anomaly response



# Requirement

- HST Automation systems shall provide the ability to respond to specific spacecraft anomalies



# Existing Monitoring Application Processors (APs)



## **MFMONTLM – Executive Telemetry Monitor Processor**

- Limit checks up to 22 telemetry items
  - Posts a status message when limits are exceeded 5 times in a row
  - Safes Payload on loss of 1 Megahertz Clock
- Actions are hardcoded
- Runs every 60 seconds

## **MFGPEF – General Purpose Event Flag Processor**

- Monitors up to 16 NSSC-1 addresses
  - Sets or clears one of 90 event flags when an action is required
  - Event Flags may be used to control a sequence of stored commanding
- Actions are controlled by a programmable table
- Runs every 0.5 second



# Approaches with No Onboard Change



- MFMONTLM – ground system takes autonomous actions based on status messages
  - Status messages potentially transient and not visible due to telemetry limitations
- MFGPEF – monitor event flags in command sequence
  - One continuously running sequence per command processor
  - Independent actions tightly coupled by single command sequence





# Approach with Onboard Change



- MFGPEF – Add optional ability to activate a command sequence
  - Make setting an Event Flag optional
  - Increase number of programmable slots to 32
  - The existing functionality would make this update fairly simple



# Requirements Creep



- Original plan to update General Purpose Events Flag processor with ability to activate stored Relative Time Command Sequences (RTCS)s deemed insufficient
  - Decision was made to add MFGPTM as a new AP in addition to MFGPEF and MFMONTLM
  - Allow multiple actions per slot



# Structural Changes for GPTM



- Event Flag Table cloned and modified
  - From 16 slots of 10 words, to 32 slots of 10 words
  - Spare fields used by Telemetry Monitor Table
    - 6 new fields added to existing 12 fields
      - New actions (Safing, RTCS, ESB, ESR), 1<sup>st</sup> Minor Frame, Track Mode
    - Modified meaning of two existing fields
      - Event flag field can disable Event Flag actions (0 now a legal value)
      - Science Instrument number expanded for use by Safing and RTCS
- Logic of Event Flag Processor modularized for Telemetry Monitor Processor
  - Divided into 5 subroutines
- Basic Design added to, but not changed



# Event Flags Table Fields



Field	Description
FREQ	Frequency to monitor Location (0 – 8191) 0.5 sec. incr.
FREQCNTR	Internal Frequency Counter from 0 to FREQ
CONSCNTR	Consecutive Counter from 0 to MINCOUNT, -1 to initialize slot
LOCATION	Address to monitor (0 – 65535)
MASK	Bit Mask ANDed with LOCATION for bits to monitor
HILIM	Higher Limit for Limit Check
LOLIM	Lower Limit for Limit Check
EVNTFLAG	Event Flag number (1 – 15, 0 is illegal value)
SI	Science Instrument number (1 - 5, or 0 for Global Flag)
CHKTYPE	Limit Check Type (0 = In Limits, 1 = Out of Limits)
ACTTYPE	Action Type (0 = Latching, 1 =Tracking)
MINCOUNT	Minimum Consecutive times limit criteria must be met before taking action (1 - 4095)



# New Fields for GPTM Table



Field	Description
SAFING	Action to Safe (0, 1). Also disables GPTM slot
RTCS	Action to activate specified RTCS (1-144, 0 = no RTCS)
ESB	Action to post Executive Status Buffer Message (0, 1)
ESR	Action to set GPTM Executive Status Report flag (0, 1)
TRKMODE	0= Continuous when action set, 1= on Transitions only
FSTMNFRM	First Minor Frame to sample location
<b>Existing fields modified:</b>	
EVNTFLAG	Event Flag number (1 – 15, 0 = no Event Flag)
SI	Science Instrument number (1 - 5, or 0 for no SI, 6 or 7 for Global Flag, 6 for ASCS Safing or RTCS, 7 for System RTCS or Payload Safing)



# GPTM Data Structure



bits	18	17	16	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1
word 1	SPARE					FREQ												
2	FREQCNTR (internal counter)																	
3	SPARE					CONSCNTR (internal counter)												
4	SPARE		LOCATION															
5	MASK																	
6	HILIM																	
7	LOLIM																	
8	<u>S</u>	<u>RTCS</u>								<u>B</u>	<u>R</u>	SI			EVNTFLAG			
9	SPARE								<u>I</u>	<u>FSTMNFRM</u>						A	C	
10	SPARE					MINCOUNT												

S - SAFING

B - ESB

R - ESR

Red = new fields for GPTM

I - TRKMODE

A - ACTTYPE

C - CHKTYPE



# Advantages of Design Reuse



- Quicker transition into development
  - Much of the desired functionality was already present
- Reuse of existing tests
- Reuse of Operational procedures and data structures
- Existing knowledge base of real world usage



# Disadvantages of Design Reuse



- Limited requirements analysis of existing functions
- Old test scripts required extensive reworking before they would run in current test environment
- Schedule assumed only originally agreed-upon modifications would be added





# Requirements Creep During Design Reviews



- Derived requirements were added to design
  - Need to stagger activation of RTCSs
    - Reduce processing load
    - Avoid nesting of one RTCS by a new RTCS
  - All slots must finish processing before next Master Timing Pulse (MTP, 500 millisecond cycle)



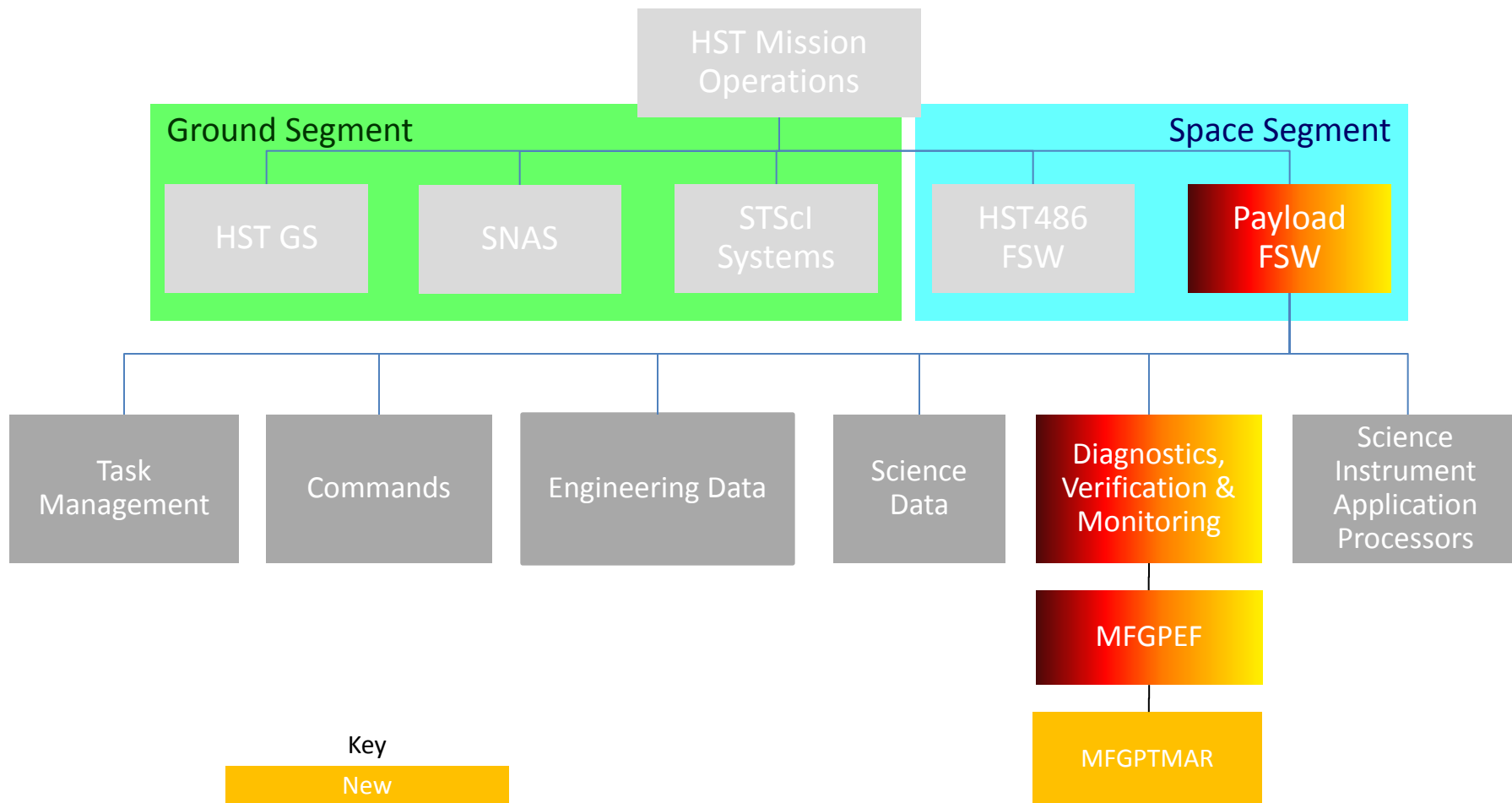
# Requirements Creep During Design Reviews



- Other requirements were slipped in as minor modifications
  - Error checking when slot initialized, and also at action time
    - Common Error Response:
      - Set ESR, Post ESB, disable GPTM Slot
        - » If Safing Initiated, or if Error initiating Safing, activating RTCS, or Setting/Resetting Event Flags
  - Prevent RTCS Nesting
    - Only activate RTCS if no other RTCS active for that SI
      - Slot will remain active if RTCS activation is delayed
  - Synchronize Monitoring to Telemetry
    - Optional first minor frame field added
      - Allow checking of subcommutated telemetry
      - Allows programming of slots for more efficient use of resources
  - Add two new tracking modes
    - Transition – only invokes RTCS or ESB on transition into action state
    - Continuous – continuously invokes RTCS or ESB while in action state



# General Purpose Telemetry Monitor Structure (initial, as modification of MFGPEF)



Key

New

Modified

No Change

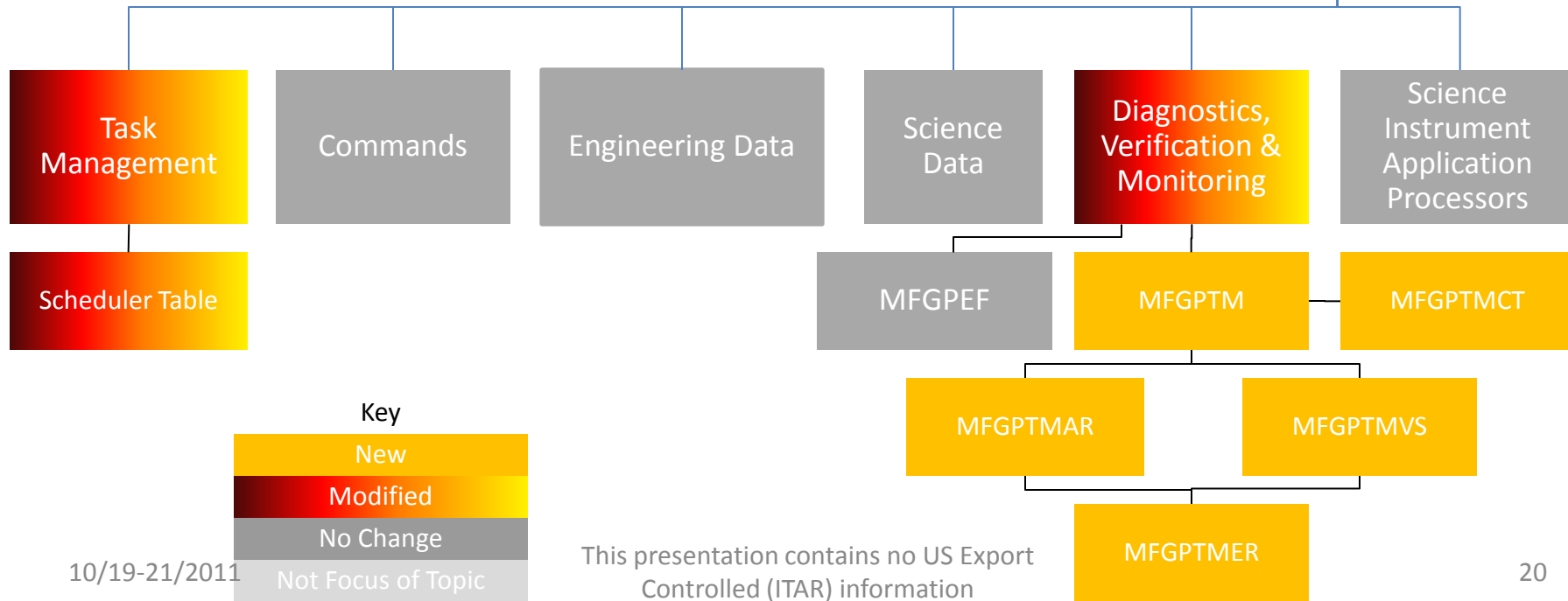
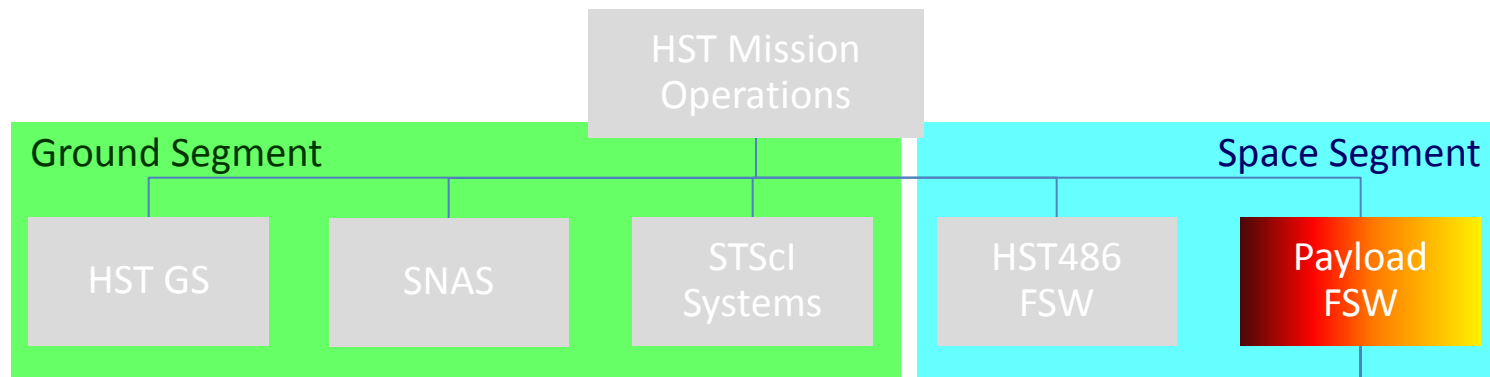
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# General Purpose Telemetry Monitor Structure (PDR)



## Key

New

Modified

No Change

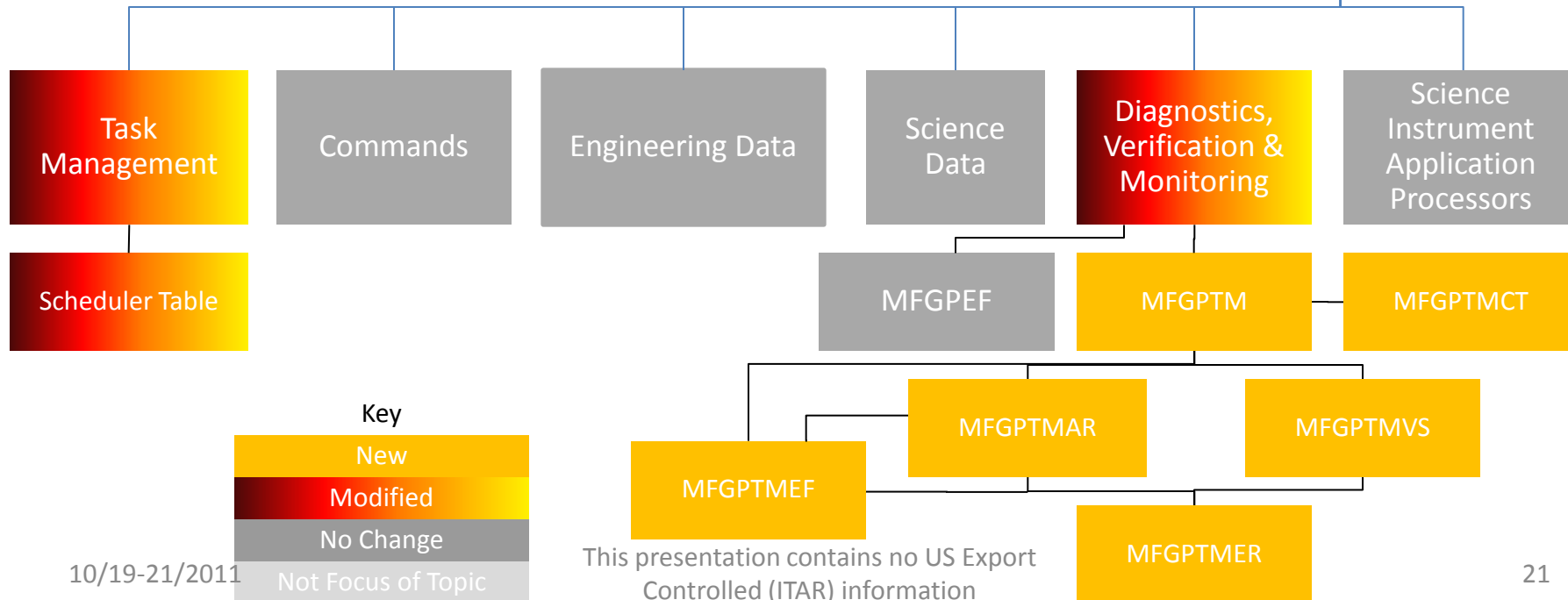
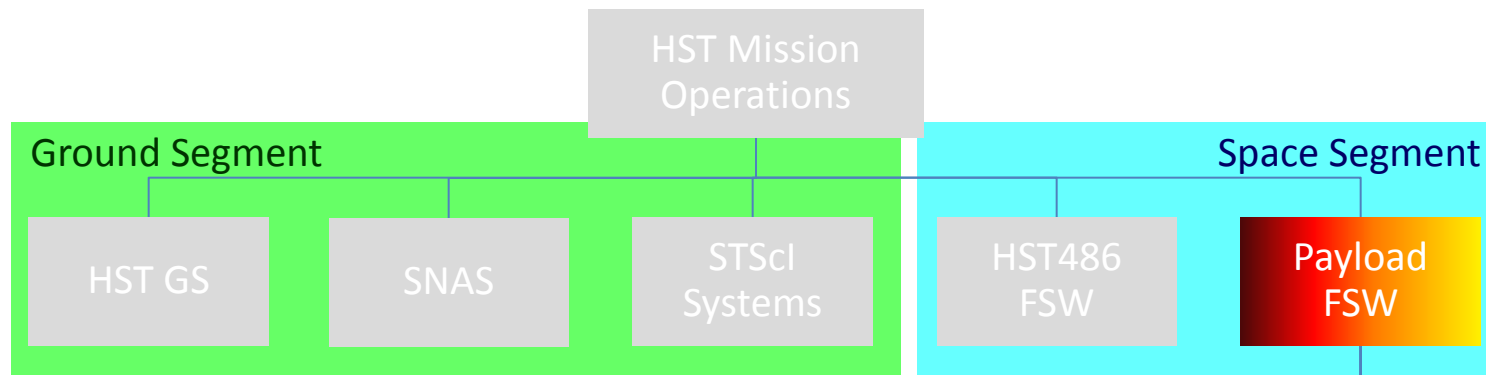
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# General Purpose Telemetry Monitor Structure (final)





# Problems due to Requirements Creep



- Modifications were made to the design without identifying the changes to the requirements
- There was no step back to re-examine the overall design in light of the changed requirements
- Unit tests matched the original/flawed design
- Design Errors slipped through as a result



# Design Errors

- Continuous tracking generated actions whether or not actions were triggered for the monitored location
  - Needed to move Event Flag code to a separate subroutine
    - Called from Main routine or Action Response subroutine
- Tracking could cause the consecutive counter to wrap
  - Caused actions to be unexpectedly triggered
  - Needed to set consecutive counter to 2 after tracking limits satisfied
- If in Tracking Mode an RTCS was not activated due to another RTCS active, the RTCS would never be activated
  - Needed to reset consecutive counter to zero after tracking limits were satisfied



# Design Error Impacts



- This was a double impact to the schedule
  - Created much more work than originally estimated
  - Flight software had to be modified and retested late in the development process
  - Symbol Of Interest (mapping of FSW variables to physical memory addresses) had already been delivered for the Project Database
    - Is a change to the SOI after delivery a big problem?





**Yes!**

Creating a new Project Database is an exacting process that requires a long lead time. Time we did not have.





# Minimizing the Impact

- Adding the MFGPTM code to the end of memory minimized the impact to the SOI
  - Order of linking assembled NSSC-1 code to create an executable image is controlled by a linker
  - GPTM was linked into end of NSSC-1 memory, with control table first
    - No symbols used operationally were affected
  - This is standard procedure, when possible, to mitigate just such a problem



# NSSC-1 Flight Software Development Process



## Team Standard

- Design
  - Write/update PDL
  - Peer review \*
- Code and Unit Test
  - Write/update Code
  - Write/update Unit Test script
  - Run Unit Test
  - Peer review code and unit test \*
- System Test
  - Write/update System Test
  - Dry run System Test
  - Peer review System Test \*
  - Formal run of System Test
- Delivery of Symbol Of Interest
- Delivery of executable

## As implemented

- Design
  - Write/update PDL
  - Peer review
    - minor updates requested
- Code and Unit Test
  - Write/update Code
  - Write/update Unit Test script
  - Run Unit Test
  - Peer review code and unit test
    - addition of new requirements change design
- System Test
  - Write/update System Test
  - Dry run System Test
  - Peer review System Test
    - Test deemed barely adequate, but no time to fix before SOI delivery
  - Formal run of System Test - passed
- Delivery of Symbol Of Interest
- More rigorous System Test – failed
- New Design and Code updates with peer reviews
- Rerun of new System Tests - passed
- Delivery of executable, and updated SOI

\* Repeat above steps as needed



# Lessons Learned



- Identify requirements changes implied by changes made during design reviews
- Review entire design in light of requirements changes



# Uses of General Purpose Telemetry Monitor



- Can replace tasks that have been done by special purpose RTCSs
- Reduces need for Ground Operations monitoring of telemetry
- GPTM is currently executing with an empty Table until HST Operations personnel develop and validate the actual slot definitions to be used on-orbit
- GPTM Programming Examples:
  - Latching example: Autonomous Safing
  - Latching example: Accommodate Yellow and Red limit actions
    - Program 2 slots to look at same telemetry point
  - Tracking example: CMOS Single Bit Error Flag Reset



# GPTM Programming

## Example 1: Safe Payload if 1 MHz Clock Lost

1	2	3	4	5	6	7	8	9	10
FREQ	FREQCNTR	CONSCNTR	LOCATION	MASK	HILIM	LOLIM	ACTION: a - SAFING b - RTCS c - ESB d - ESR e - SI f - EVNTFLAG	TYPEFMF: a - TRKMODE b - FSTMNFRM c - ACTTYPE d - CHKTYPE	MINCOUNT
1  Every 0.5 sec	0777777 (-1)  initialize slot	0  always load zero	NSSC-1 address of 1 No 1 MHz Clock flag telemetry word	000002  Monitor bit 2  No 1 MHz Clock flag	000000  High limit = 0	000000  Low limit = 0	0400160  a=1 - Safing b=0 - NoRTCS c=0 - NoPost d=0 - NoSet e=7 - PAYLOAD f=0 - NoEF	000001  a=0 Continuous b=0 1 <sup>st</sup> mf c=0 Latching d=1 OutLimit	5  Take action on 5 <sup>th</sup> consecutive out of limit condition

- Initialize Payload Safing when “No 1 MHz Clock” flag is set 5 consecutive times
- Due to Safing, FREQ will be set to 0, disabling the slot until it is reset

Note: Values listed are “for example only”.





# GPTM Programming

## Example 2: Turn Off SDF if Side B Temperature is Out of Limits



Slot	1 FREQ	2 FREQCNTR	3 CONSCNTR	4 LOCATION	5 MASK	6 HILIM	7 LOLIM	8 ACTION: a - SAFING b - RTCS c - ESB d - ESR e - SI f - EVNTFLAG	9 TYPEFMF: a - TRKMODE b - FSTMNFRM c - ACTTYPE d - CHKTYPE	10 MINCOUNT
n+1	024 (20)  Every 10 sec.	0777777 (-1)  Initialize slot	0  Always load 0	NSSC-1 address of SDF Side B Temperature telemetry	000377  monitor 8 low order bits	000342  Red Upper Limit	000052  Red Lower Limit	0217760 a=0 - NoSafe b=143 - RTCS TOFFSDF c=1 - Post d=1 - Set e=7 - SYSTEM f=0 - NoEF	000025  a=0 - Continuous b=4 - 1 <sup>st</sup> mf c=0 - Latching d=1 - OutLimit	3  Take action on 3 <sup>rd</sup> consecutive out of limits
n+2	024 (20)  Every 10 sec.	0777777 (-1)  Initialize slot	0  Always load 0	NSSC-1 address of SDF Side B Temperature telemetry	000377  Monitor 8 low order bits	000322  Yellow Upper Limit	000060  Yellow Lower Limit	000560 a=0 - NoSafe b=0 - NoRTCS c=1 - Post d=0 - NoSet e=7 - SYSTEM f=0 - NoEF	000025  a=0 - Continuous b=4 - 1 <sup>st</sup> mf c=0 - Latching d=1 - OutLimit	3  Take action on 3 <sup>rd</sup> consecutive out of limits

- Programming two slots to monitor the same telemetry allows red and yellow limit actions
  - Slot n+2 Post an ESB only
  - Slot n+1 activates RTCS 143 to Turn Off SDF, sets ESR flag and Posts an ESB
- Due to Latching, FREQ will be set to 0, disabling the triggered slot until it is reset

**Note: Values listed are “for example only”.**



# GPTM Programming

## Example 3: Reset CMOS Single Bit Error Flag when Set



1	2	3	4	5	6	7	8	9	10
FREQ	FREQCNTR	CONSCNTR	LOCATION	MASK	HILIM (EU)	LOLIM (EU)	ACTION: a - SAFING b - RTCS c - ESB d - ESR e - SI f - EVNTFLAG	TYPEFMF: a - TRKMODE b - FSTMNFRM c - ACTTYPE d - CHKTYPE	MINCOUNT
024 (20)  Every 10 sec.	0777777 (-1)  Initialize slot	0  Always load 0	NSSC-1 address of CMOS Single Bit Error Flag telemetry word	000040  Monitor CMOS Single Bit Error Flag	000000  Upper Limit	000000  Lower Limit	0220160  a=0 - NoSafing b=144 - RTCS CCMOSSBE c=0 - NoPost d=0 - NoSet e=7- SYSTEM f=0 - NoEF	001037  a=1 - Transition b=7 - 1 <sup>st</sup> mf c=1 - Tracking d=1 - OutLimit	0  Not used for tracking

- When there is a Single Bit Error in NSSC-1 CMOS memory, the built-in EDAC function corrects the error and sets a flag in telemetry
- This GPTM slot will activate RTCS 144 to clear the SBE flag when it is set
  - Action Type = Tracking – the SBE telemetry is always monitored
  - Track Mode = Transition – the action takes place just when the SBE flag is set

**Note:** Values listed are “for example only”.





# Conclusion

- Design reuse has distinct advantages
  - Design is already tested and understood
  - A user knowledge base already exists
  - Test scripts already exist
- A reused design should be treated as new in terms of requirements analysis, and testing
  - Minor changes can have unexpected design implications
- Design for future maintainability



That's a wrap.



Detail of one of the simulator boards in our NSSC-1 FSW Lab.



# BACKUP SLIDES



# Acronyms



**ACS** – Advanced Camera for Surveys (SI 1)

**AP** – Application Processor

**BCU** – Bus Coupler Unit

**C&DH** – Command & Data Handling

**COS** – Cosmic Origins Spectrograph (SI 4)

**CPM** – Central Processor Module

**CU** – Control Unit

**ESB** – Executive Status Buffer message

**ESR** – Executive Status Report flag

**FSW** – Flight Software

**GPTM** – General Purpose Telemetry Monitor

**HST** – Hubble Space Telescope

**MOSES** – Mission Operations, System Engineering & Software

**NASA** – National Aeronautics and Space Administration

**NICMOS** – Near Infrared Camera and

Multi-Object Spectrometer (SI 2)

**NSSC-1** – NASA Standard Spacecraft Computer, Model 1

**PCU** – Power Control Unit

**PDL** – Program Design Language

**RIU** – Remote Interface Unit

**RM** – Remote Module

**RTCS** – Relative Time Command Sequence

**SAA** – South Atlantic Anomaly

**SI** – Science Instrument

**SDF** – Science Data Formatter

**SMM** – Solar Maximum Mission

**SSM** – Support Systems Module

**STINT** – Standard Interface

**STIS** – Space Telescope Imaging Spectrograph (SI 3)

**WFC3** – Wide Field Camera 3 (SI 5)





# GPTM Code Modules



- MFGPTM, General Purpose Telemetry Monitor (New)
  - General Purpose Telemetry Monitor (GPTM) Application Processor
  - 569 Lines of code
- MFGPTMAR, GPTM Action Response (New)
  - Handles any action programmed into a GPTM control table slot
  - 294 Lines of code = 259 Lines in module + 35 Lines in calling macro
- MFGPTMCT, GPTM Control Table (New)
  - Contains the 32 slot programmable GPTM Control Table
    - GPTM Control Table will be delivered empty
  - 320 word table - 10 words per slot
- MFGPTMEF, GPTM Event Flag (New)
  - Controls the setting and clearing of event flags by the GPTM AP
  - 192 Lines of code = 157 Lines in module + 35 lines in calling macro
- MFGPTMVS, GPTM Validate Slot (New)
  - Examines GPTM Control Table slots
  - Disables a slot found to have coding errors
  - 320 Lines of code
- MFGPTMER, GPTM Error Response (New)
  - Posts an ESB message with a parameter containing an error return code in bits 10-7, and GPTM Control Table slot number in bits 6-1
  - 114 Lines of code = 75 Lines in module + 39 Lines in calling macro
- MFUNIQUE, Mission Unique Data (Modified)
  - Add Executive Processor 9, General Purpose Telemetry Monitor, to the Scheduler Table
  - 1405 Lines of existing code, 11 lines modified